NTSB ID: NYC04LA035	Aircraft Registration Number: N999GC							
Occurrence Date: 11/16/2003	Most Critical Injury: Fatal							
Occurrence Type: Accident	Investigated By: NTSB							

Nearest City/Place	State	Zip Code	Local Time	Time Zone	
Nelsonia	VA	23414	1119	EST	
Airport Proximity: Off Airport/Airstrip	Distance From	m Landing Facility:	•	Direction Fro	m Airport:

### Aircraft Information Summary

Aircraft Manufacturer	Model/Series	Type of Aircraft
Northam	RAF 2000	Gyrocraft

# Sightseeing Flight: No

# Air Medical Transport Flight: No

## Narrative

Brief narrative statement of facts, conditions and circumstances pertinent to the accident/incident:

On November 16, 2003, at 1119 eastern standard time, a homebuilt RAF 2000 gyroplane, N999GC, was destroyed when it struck the ground in Nelsonia, Virginia. The certificated airline transport pilot was fatally injured. Visual meteorological conditions prevailed for the personal flight that originated from Accomack County Airport (MFV), Melfa, Virginia. No flight plan had been filed for the local flight that was conducted under 14 CFR Part 91.

Personnel at the departure airport reported that the pilot had initiated a takeoff, and then aborted the takeoff for unknown reasons. He pulled clear of the runway and stayed there for about 10 minutes with the engine idling. He then pulled back on the runway and departed about 1028. That was the last he was seen at the airport.

The gyroplane was next observed by a witness, who was located about 2 miles northwest of the accident site. The witness reported that when she first observed the gyroplane, the nose was oscillating up and down. The oscillations smoothed out, and the gyroplane continued in straight and level flight to the southeast. Shortly after the gyroplane disappeared from view, she observed a rising column of black smoke coming from the direction the gyroplane was headed.

Additional interviews of witnesses in the area revealed that three of the witnesses reported that the gyroplane was in level flight when it rolled left to the inverted position, and then descended in a nose down attitude and impacted the ground. One other witness thought that the gyroplane had first pitched up to a near vertical attitude, and then reversed course and impacted the ground in a nose down attitude. The witnesses agreed that the gyroplane burst into flames after the impact.

The examination of the wreckage revealed that both main rotor blades had red paint transfer marks on the top and bottom side of the blades, 67 inches from the end of the blade straps, consistent with the location of the vertical stabilizer. In addition, there were strike marks on underside of both blade tension straps, consistent with the location of the engine propeller.

One main rotor blade was straight, while the other was bent up about 20 degrees at the mid-span location. There was no evidence of rearward bending of the blades.

A portion of the rudder was consumed in the fire. However, the upper portion of the rudder was separated from the aircraft and found outside of the burn area. There was impact damage on the left side of the separated piece.

The gyroplane was equipped with a fixed incident, horizontal stabilizer, which had winglets that were canted 20 degrees outboard of vertical. The condition of the right winglet was consistent with a main rotor blade strike.

Flight control continuity was not confirmed due to impact and fire damage.

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Narrative (Continued)

In addition, the main mast was separated below the rotor hub, and one propeller blade was found separated from the remainder of the blade.

Re-examination of the impact site revealed that all items found away from the main impact area were light items and subject to lateral movement from wind. The direction of movement of the objects did not match the flight path described by witnesses.

The gyroplane had been modified with the addition of a fixed incident horizontal stabilizer. This was not recommended by the kit manufacturer. However, the experimental airworthiness certificate of the gyroplane did not preclude additions, deletions, or modifications from the original kit design.

The pilot was reported to have accumulated about 70 hours in make and model. This included 20 hours of dual instruction in another RAF 2000, and then 50 hours in his own RAF 2000.

The pilot's flight experience was reported to be in excess of 14,550 hours. He held ratings for airplanes single and multi-engine, and instrument airplane. He did not posses a rotorcraft, gyroplane, category and class rating, nor was he required under existing rules from the Federal Aviation Administration (FAA).

Toxicological testing conducted by the FAA Toxicology Accident Research Laboratory, Oklahoma City, Oklahoma, revealed the following:

17 mg/dl ethanol detected in urine
2 mg/dl acetaldehyde detected in urine
2 mg/dl n-propanol detected in urine
86.36 ug/ml acetaminophen detected in urine
452 ug/ml salicylate detected in urine
No ethanol in lung
No ethanol in brain

The toxicological report also noted that the samples were received in putrefied condition.

The toxicological report received from the State of Virginia reported an alcohol content of 0.01 percent ethanol by volume. In addition, ethanol was not detected in vitreous fluid.

On November 18, 2003, an autopsy was conducted by Wendy Gunther, MD, Assistant Chief Medical Examiner, Tidewater District, State of Virginia.

According to the FAA-H-8083-21 Rotorcraft Flying Handbook, Chapter 21, Gyroplane Emergencies, pilot induced oscillations (PIO) can occur in both the longitudinal and lateral axis. The publication further stated:

- "...As with most other rotor-wing aircraft, gyroplanes experience a slight delay between control input and the reaction of the aircraft. This delay may cause an inexperienced pilot to apply more control input than required, causing a greater aircraft response than was desired. Once the error has been recognized, opposite control input is applied to correct the flight attitude. Because of the nature of the delay in aircraft response, it is possible for the corrections to be out of synchronization with the movements of the aircraft and aggravate the undesired changes in attitude. The result is PIO, or unintentional oscillations that can grow rapidly in magnitude...."
- "...the stability of a gyroplane is greatly influenced by rotor force. If rotor force is rapidly removed, some gyroplanes have a tendency to pitch forward abruptly. This is often referred to as a forward tumble, buntover, or power pushover. Removing the rotor force is often referred to as

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Narrative (Continued)

unloading the rotor, and can occur if pilot-induced oscillations become excessive, if extremely turbulent conditions are encountered, or the nose of the gyroplane is pushed forward rapidly after a steep climb."

- "A power pushover can occur on some gyroplanes that have the propeller thrust line above the center of gravity and do not have an adequate horizontal stabilizer. In this case, when the rotor is unloaded, the propeller thrust magnifies the pitching moment around the center of gravity. Unless a correction is made, this nose pitching action could become self-sustaining and irreversible. An adequate horizontal stabilizer slows the pitching rate and allows time for recovery."
- "Since there is some disagreement between manufacturers as to the proper recovery procedure for this situation, you must check with the manufacturer of your gyroplane. In most cases, you need to remove power and load the rotor blades. Some manufacturers, especially those with gyroplanes where the propeller thrust line is above the center of gravity, recommend that you need to immediately remove power in order to prevent a power pushover situation. Other manufacturers recommend that you first try to load the rotor blades. For the proper positioning of the cyclic when loading up the rotor blades, check with the manufacturer."
- "When compared to other aircraft, the gyroplane is just as safe and very reliable. The most important factor, as in all aircraft, is pilot proficiency. Proper training and flight experience helps prevent the risks associated with pilot-induced oscillation or buntover...."

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AVIATION	Occurren	nce Type:	: Accident									
Landing Facility/Approach Information												
Airport Name	Airp	oort ID:	Airport Elevat	rport Elevation Runway Used			Runwa	y Length	Run	way Width		
			Ft.	. MSL	NA							
Runway Surface Type: Unknown												
Runway Surface Condition: Unknown												
- · · · NONE												
Type Instrument Approach: NONE												
VFR Approach/Landing: None												
Aircraft Information												
Aircraft Manufacturer		Model/						Serial N				
Northam		RAF 2	2000					H2-02	-13540			
Airworthiness Certificate(s): Experimental (Special)												
Landing Gear Type: Tricycle												
Homebuilt Aircraft? Yes Number of Seats: 2			d Max Gross W	1540 LBS Number			of Engine					
Engine Type: Reciprocating		ngine Mai Subaru	nufacturer:			Model/Se EJ 22	ries:			ed Power: 0 HP		
- Aircraft Inspection Information												
Type of Last Inspection	Dat	te of Last	t Inspection	Ti	ime Sir	nce Last Inspe	ection	1	Airframe Total Time			
Conditional	07	7/07/200	)3				50 Ho		50 Hours			
- Emergency Locator Transmitter (ELT) Information												
ELT Installed? No ELT Operate	ed?			ELT A	اded ir	n Locating Ac	cident S	ite?				
Owner/Operator Information												
Registered Aircraft Owner		Street Address										
Richard A. Northam		City State										
			VA	23405								
Operator of Aircraft		Street Address Same as Reg'd Aircraft Owner										
Same as Reg'd Aircraft Owner	-	City Same as Regularitationnel								Zip Code		
Same as Neg a Ancian Owner										-		
Operator Does Business As:					Op	perator Design	nator Co	de:				
- Type of U.S. Certificate(s) Held: None												
Air Carrier Operating Certificate(s):												
Operating Certificate:			Operator C	Certificat	te:							
Regulation Flight Conducted Under: Part 91: Genera	al Aviation											
Type of Flight Operation Conducted: Personal												
Type of Flight operation conducted. Foreona.										1		

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	AVIATI	Occurrence Type: Accident												
First Pilot	Information													
Name		City	Sta					te C	ate of Birth	Age				
Richard A. Northam Mach								0			VA		On File	51
Sex: M	Seat Occupied	: Left	Prir	ncipal Profes	sion: Civilia	n Pilot				Cei	tificat	e Numbe	er: On File	
Certificate(s): Airline Transport; Flight Instructor; Commercial; Flight Engineer														
Airplane Ra	Airplane Rating(s): Multi-engine Land; Single-engine Land													
Rotorcraft/Glider/LTA: None														
Instrument Rating(s): Airplane														
Instructor Rating(s): None														
Type Rating/Endorsement for Accident/Incident Aircraft? No Current Biennial Flight Review? 03/19/2003														
Medical Cer	rt.: Class 1	Medica	al Cert. Status	S: Valid Me	dicalw/ wa	aivers/li	im.			Date of La	ast Me	edical Ex	am: 09/29/20	003
I														
- Flight Tim	e Matrix	All A/C	This Make and Model	Airplane Single Engine	Airplane Mult-Engine	Night		Actual	Instrument simulated			Rotorcraft	Glider	Lighter Than Air
Total Time		14550	70											
Pilot In Com	nmand(PIC)		70								$\perp$			
Instructor											$\perp$			
Last 90 Day						-					_			
Last 30 Day						+					+			
Last 24 Hou						<del> </del>	<b>T</b>	D.		10. \			and Bilato M	
Seatbelt Us	ed? Yes	Shou	llder Harness	Used? Yes			IOXICC	Diogy Pe	errorme	ed? Yes		Sec	cond Pilot? No	)
Flight Pla	n/Itinerary													
	ht Plan Filed: No	one												
Departure F	Point						State		Airpor	t Identifie	r	Depart	Time Zone	
Melfa							VA		MFV			1028		EST
Destination							State		Airport Identifier					
Local Flig	ht						1		MFV					
Type of Cle	arance: None													
Type of Airs	space: Class	G												
Weather	Information													
Source of E	-	ord of briefi	ng											
Method of E	Briefing: Unkno	wn												
				FACTUAI	REPORT	- AVIA	ATION	٧						Page 3

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	ETYBOR		Occurrence Type: Accident										
Weather Information													
WOF ID	Observation Time	Time Zone	W	OF Elevati	on	WOF D	Distance Fr	rom Acci	dent Site	n Accident Site	Э		
WAL	1054	EST		40 Ft.	MSL				10 NM			57 Deg.	Mag.
Sky/Lowes	st Cloud Condition:						Ft.	AGL	Condition o	f Ligh	nt: Day		
Lowest Ce	iling: Overcast			5000 Ft.	AGL	Visik	oility:	10	SM	Altir	meter:	30.23	"Hg
Temperatu	ure: 14 °C	Dew Point:		6 °C	Wind	Direction	ո։ 180			Der	nsity Altitude:	-355	Ft.
Wind Spee	ed: 9	Gusts:			Weath	ner Cond	tions at Ac	ccident S	Site: Visual C	Condi	itions		
Visibility (R	RVR): Ft.	Visibility	(RVV)		SM	Intensi	ty of Preci	pitation:					
Restriction	s to Visibility: None												
Type of Pre	ecipitation: None												
	•												
Accident	Information												
Aircraft Dar				Aircraft Fire	e:				Aircraft Exp	losio	n		
Classificati	on:												
- Injury Su	mmary Matrix	Fatal	Serious	Mino	or	None	TOTAL						
First Pil	lot	1						1					
Second	d Pilot							$\neg$					
Studen	ut Pilot						1	$\neg$					
Flight II	nstructor							$\neg$					
Check I	Pilot							$\neg$					
Flight E	Engineer												
Cabin /	Attendants												
Other C	Crew							$\neg$					
Passen	ngers						†	$\neg$					
- TOTAL A	ABOARD -	1						1					
Other G	Ground			$\top$				$\neg$					
- GRANE	O TOTAL -	1						1					

National Transportation Safety Board

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Investigator-In-Charge (IIC)

Robert L. Hancock

Additional Persons Participating in This Accident/Incident Investigation:

Arthur Munns Aviation Safety Inspector Federal Aviation Administration Richmond, VA